



# IN-SPACE WATER SUPPLY CHAIN SERVICING THE U.S. MILITARY

*A PRELIMINARY ESTIMATE OF FUTURE  
POTENTIAL U.S. MILITARY SUPPLY AND  
DEMAND FOR IN-SPACE WATER-BASED FUEL*

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SPACE RESOURCES ROUNDTABLE - JUNE 2019

# WATER SUPPLY AND DEMAND

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## Supply

	Moon	Near Earth Asteroids	Asteroid Belt
Pros	Close	Potential low delta-V	Enormous supply
	Diverse resources & applications	Significant supply	Diverse resources
Cons	Relatively high delta-V	Highly variable	High delta-V
	Finite supply	Significant unknowns	Far away

## Demand

- Demand estimates based on current U.S. military assets in orbit
- Future trends for satellite development are unknown
- Future assets may be smaller & decentralized, but advancements could dramatically change this

# CALCULATING DEMAND

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# DEMAND ASSUMPTIONS AND METHODOLOGY

## Assumptions

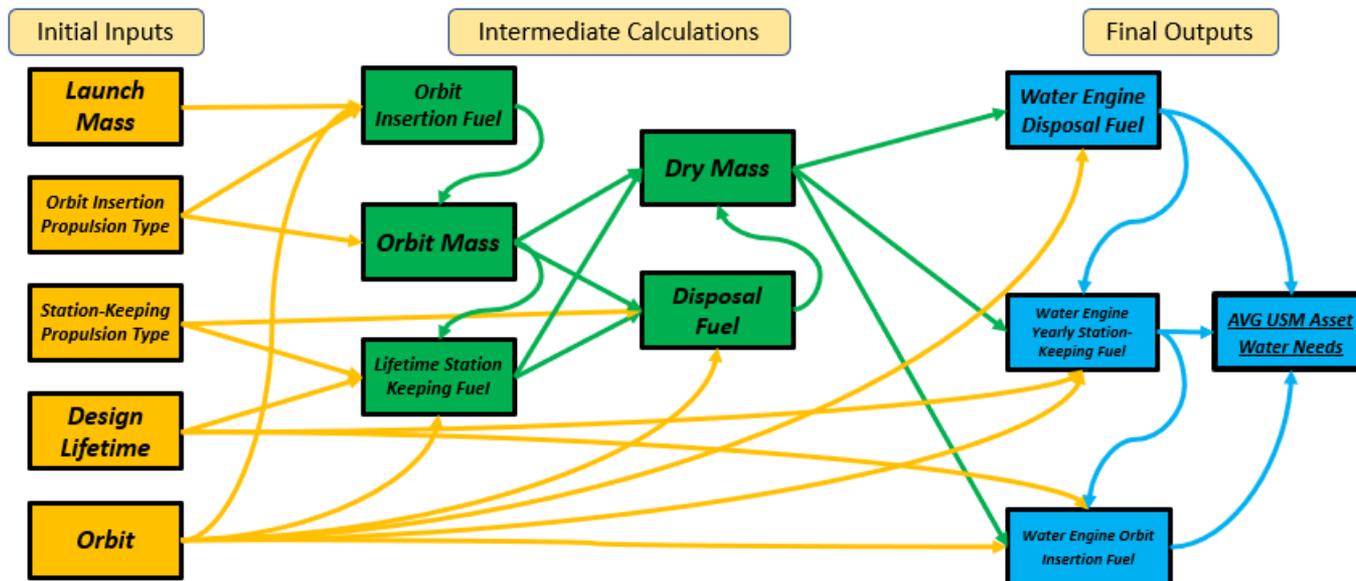
Deriving water needs:

- All fuel is replaced with water (ISP ~180)
- Fueling occurs in LEO for deployment
- Dry mass of satellites does not change
- CubeSats not considered

Future assets:

- Follow same orbital distribution
- Lifetime of 20 years
- 10X current maneuverability
- 10% share of all ISRU water

## Methodology



# USG ASSETS RESULTS

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Based on ~130 military assets, it was estimated:

- Current demand is ~45 tons of water per year
- 333 kg of water per asset per year

A future water propelled U.S. military asset would require:

- 3,000 kg for deployment
- 130 kg for disposal
- 610 kg per year for station keeping

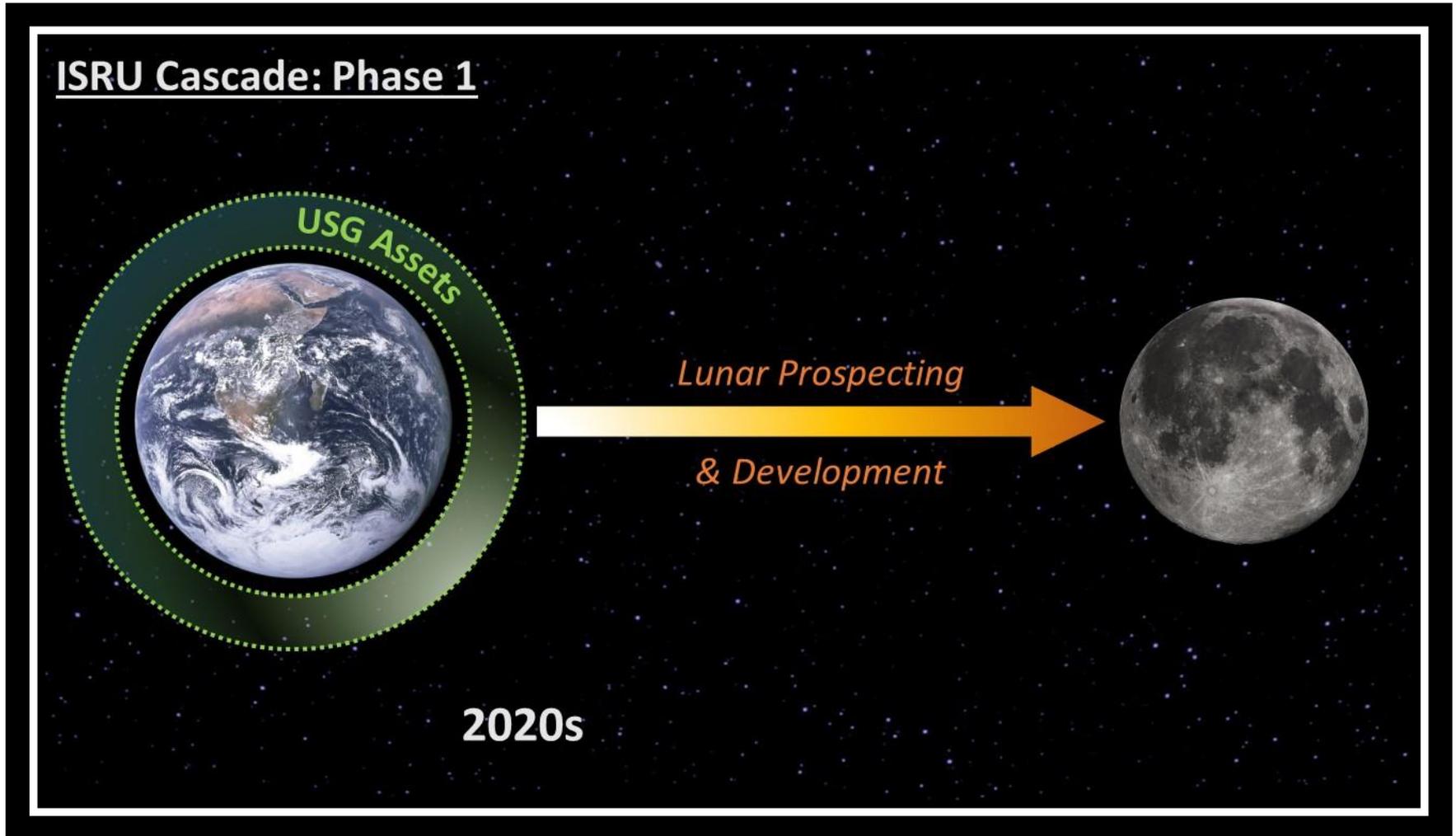
Supply chain demand:

- ~40% Fuel for LEO
- ~20% Fuel for MEO
- ~40% Fuel for GEO

Transportation to destination orbit needs 50% for LH2/LOX and 10% for ion/plasma

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# SUPPLY CHAIN ROAD MAP

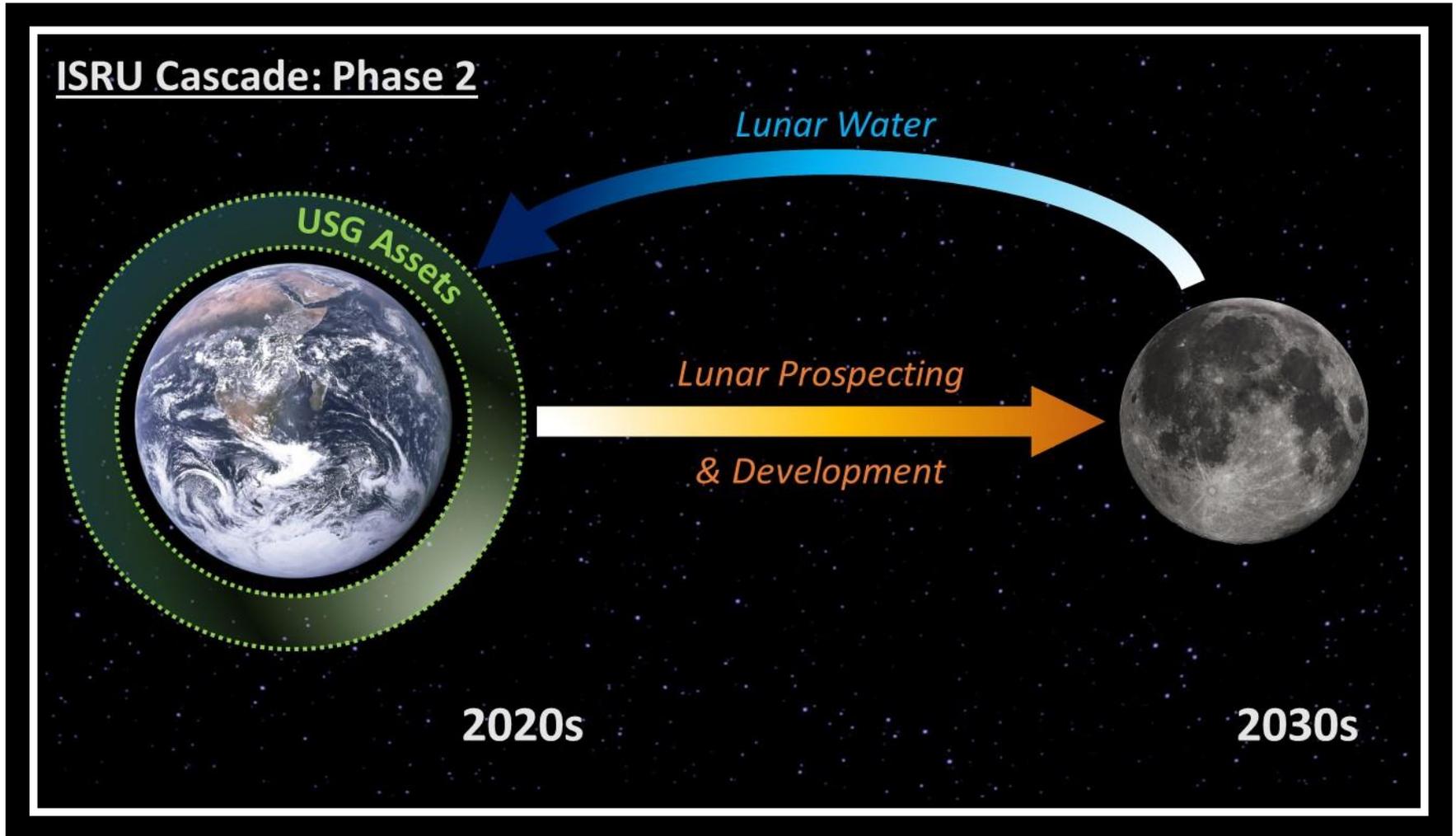


# CALCULATING SUPPLY 1: THE MOON

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# SUPPLY CHAIN ROAD MAP



# LUNAR ASSUMPTIONS AND METHODS

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## Assumptions

- $1.2 \times 10^{12}$  kg of water located at the Lunar Poles
- U.S. Military has 10% share of total
- About 2/3 of the dry mass of Lunar escape vehicles is payload
- LH2/LOX fuel for Lunar escape
- Ion/plasma propulsion used for final fuel delivery

## Methods

- 10% multiplier for 10% share of total water
- ~50% multiplier for water losses due to Lunar escape
- ~90% multiplier for water losses due to destination orbit insertion

**Leaving  $5.2 \times 10^{10}$  kg for military assets**

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# LUNAR RESULTS

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Accessible with  
conventional technology

## Independent Lunar Supply Chain

Relatively high  
magnitudes of water

**Source Water Mass**

$1.2 \times 10^{12}$  kg

Important stepping  
stone to future supply  
chains

**Delivered Water Mass**

$5.2 \times 10^{10}$  kg

Easier to disrupt

**Theoretical Number of  
Deployable Assets**

**~3.4 Million Assets**

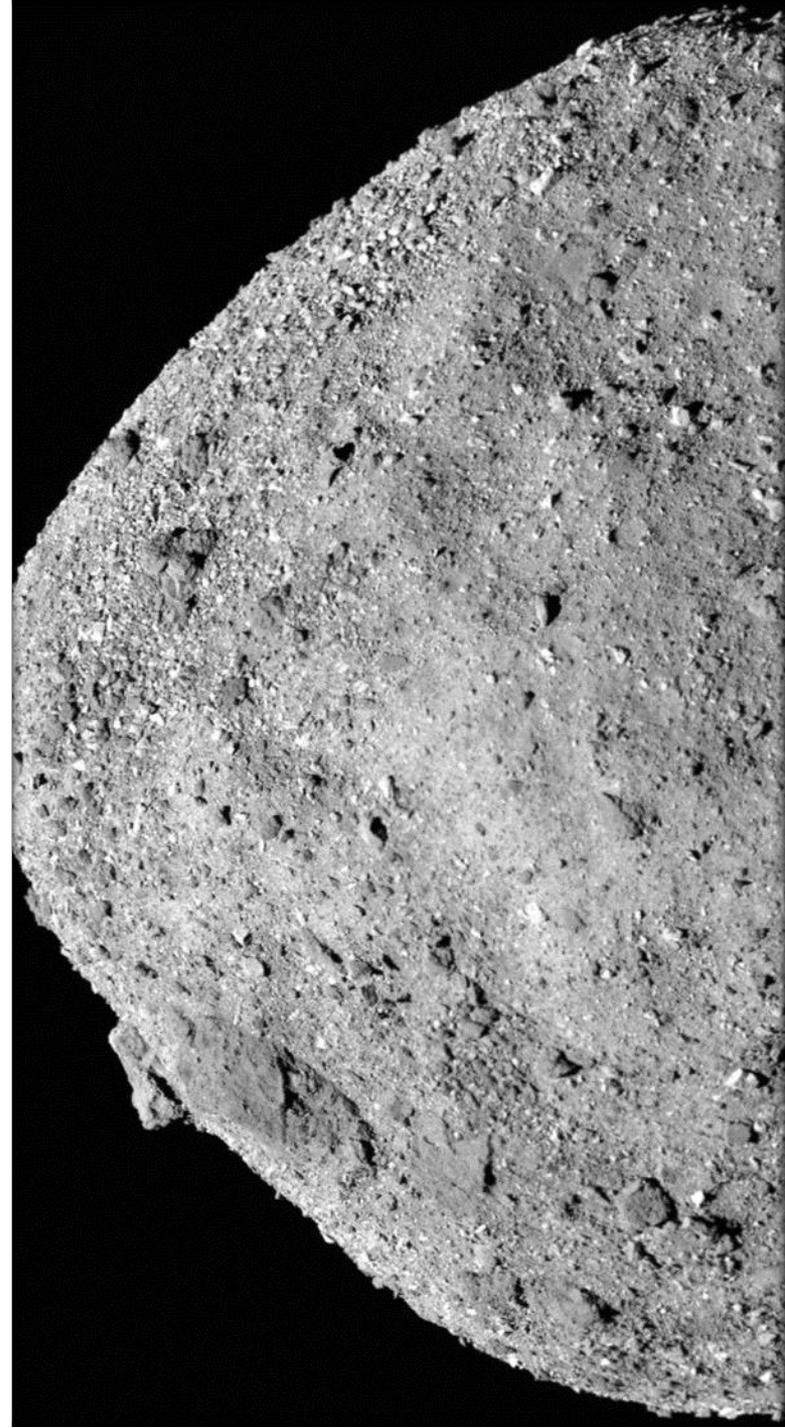
More Competition

**Time Until Depletion**

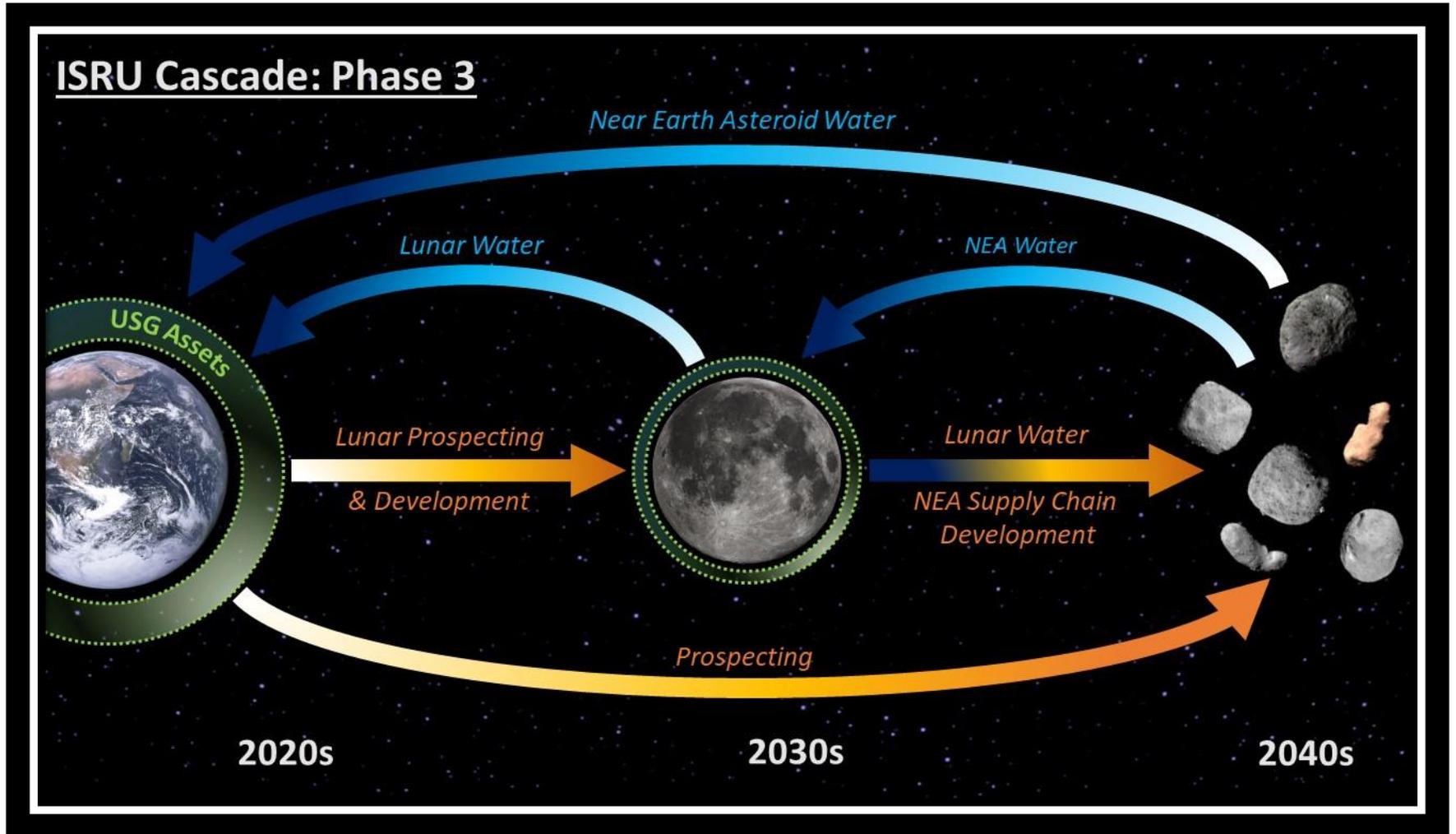
**~510 years**

# CALCULATING SUPPLY 2: NEAR EARTH ASTEROIDS

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# SUPPLY CHAIN ROAD MAP



# NEAR TERM NEA ASSUMPTIONS AND METHODS

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## Assumptions

- Only 5-30m asteroids are minable
- C-types make up 20% of NEAs
- Even distribution of asteroid size and type
- U.S. Military has a 10% share
- Solar baking method is scalable
- Ratio of asteroid to spacecraft is 60
- Min of 50% of water reserved for return
- 75% of the water is mined out and the remainder of the asteroid is ditched
- Fuel to despin asteroids is negligible
- LH2/LOX is used for outbound
- Derived LH2/LOX fuel for Earth return
- Ion/plasma propulsion is used for fuel delivery

## Methods

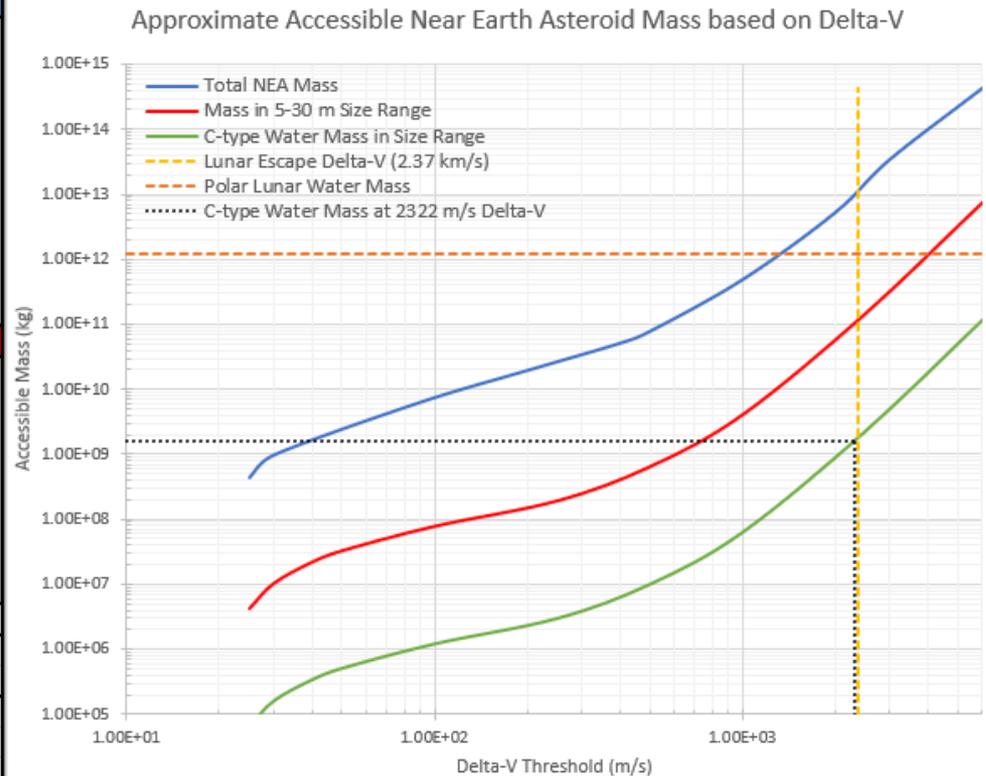
- Use assumptions to find delta-V max
- Use delta-V in broken plane delta-V function to estimate mass in range
- Adjust based on ratio in size range
- 20% multiplier for C-types
- Density adjustment multiplier
- 10% multiplier for percent water
- 75% multiplier for water taken
- 50% multiplier for water saved
- Subtract water inbound for net
- Adjust using Reiman sums to account for greater recovery at all lower delta-Vs
- 10% share of total water
- ~90% multiplier for water loses

**Leaving  $5.3 \times 10^7$  kg for Military Assets**

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# NEAR TERM NEA MODEL

Asteroid Retrieval via Bagging and Water Thruster		
Initial Inputs		
Minimum Asteroid Diameter (m):	5	
Maximum Asteroid Diameter (m):	30	
Asteroid to Dry Spacecraft Mass Ratio:	60	
% of Asteroid's Water Saved for Recovery:	50	
Specific Impulse to Asteroid (s):	450	
Specific Impulse back from Asteroid (s):	450	
Ditch Rest of Asteroid? (1=No, 2=Yes):	2	
% of Asteroid H2O Baked out for Recovery:	75	
Outputs		
Max Delta-V (m/s):	2,321.73	
Water Mass from C-types in Range (kg):	1.61E+09	
Water Mass Recoverable w/ Settings (kg):	6.93E+08	
Percentage of Water Mass Recoverable:	43.01	
Net Water Mass Gained w/ Settings (kg):	5.83E+08	
Overall Net Water Return Ratio	5.28	
Number of C-type Asteroids in Range:	15,262	
Test Cases at Max Delta-V:		
	5 m (min)	30 m (max)
Asteroid Mass (kg):	1.31E+05	2.83E+07
Spacecraft Mass (kg):	2,181.66	471,238.90
Recovered Water Mass (kg):	4,908.74	1,060,287.52
Initial Water Needed (kg):	1,510.38	326,242.31
Water Return Ratio:	3.25	3.25



# NEAR TERM NEA RESULTS

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Comparatively low water mass

Extremely efficient for early ISRU cascade all within Lunar delta-V

Water return ratio here is 5 times the investment

Provides value as a redundant supply chain

Hard to disrupt

<b>Independent Near Term NEA Supply Chain</b>	
<b>Source Water Mass</b>	1.6 x 10 <sup>9</sup> kg
<b>Delivered Water Mass</b>	5.3 x 10 <sup>7</sup> kg
<b>Theoretical Number of Deployable Assets</b>	<b>~3400 Assets</b>
<b>Time Until Depletion</b>	<b>~170 years</b>

# LONG TERM NEA ASSUMPTIONS AND METHODS

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## Assumptions

- All NEAs accessible with minimal in-bound loss
- C-types make up 40% of NEAs
- Even distribution of asteroid size and type across delta-V ranges
- All C-type asteroids can be mined
- 100% of the water is mined and the remainder of the asteroid is ditched
- Fuel to despin asteroids is negligible
- U.S. Military has 10% share of water
- Derived LH2/LOX fuel for Earth return
- Ion/plasma propulsion used for final fuel delivery

## Methods

- 20% multiplier for C-types
- Density adjustment multiplier
- 10% multiplier for percent Water
- 75% multiplier for water taken
- ~16% multiplier for losses due to Earth return
- 10% multiplier for military share
- ~90% multiplier for water losses due to destination orbit insertion

**Leaving  $9.8 \times 10^{12}$  kg for Military Assets**

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# LONG TERM NEA RESULTS

Tremendous resources available in the NEA population

Much of it is currently inaccessible

Requires significant R&D but it's worthwhile

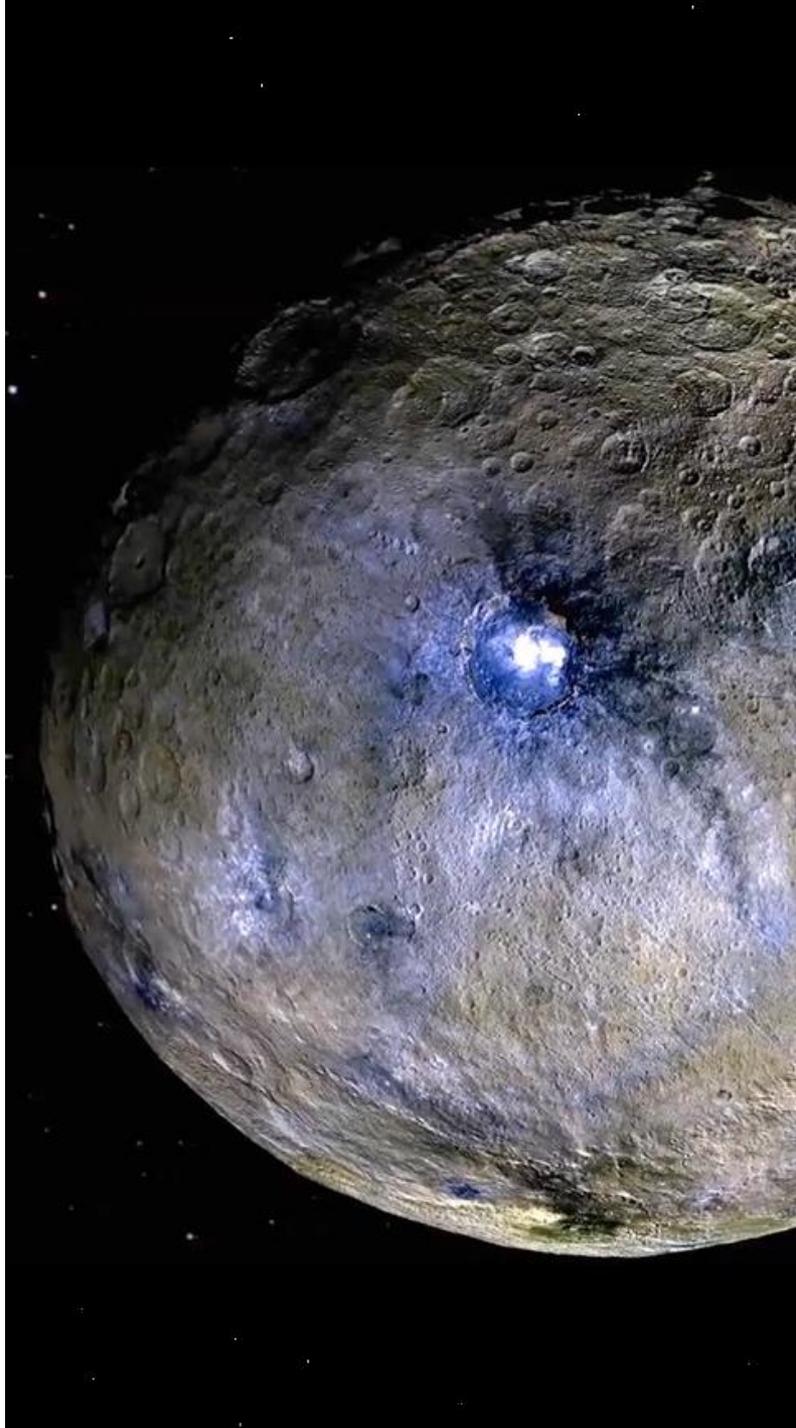
Hard to disrupt

Independent Long Term NEA Supply Chain	
Source Water Mass	6.7 x 10 <sup>14</sup> kg
Delivered Water Mass	9.8 x 10 <sup>12</sup> kg
Theoretical Number of Deployable Assets	1 Million Assets*
Time Until Depletion	12,800 years*

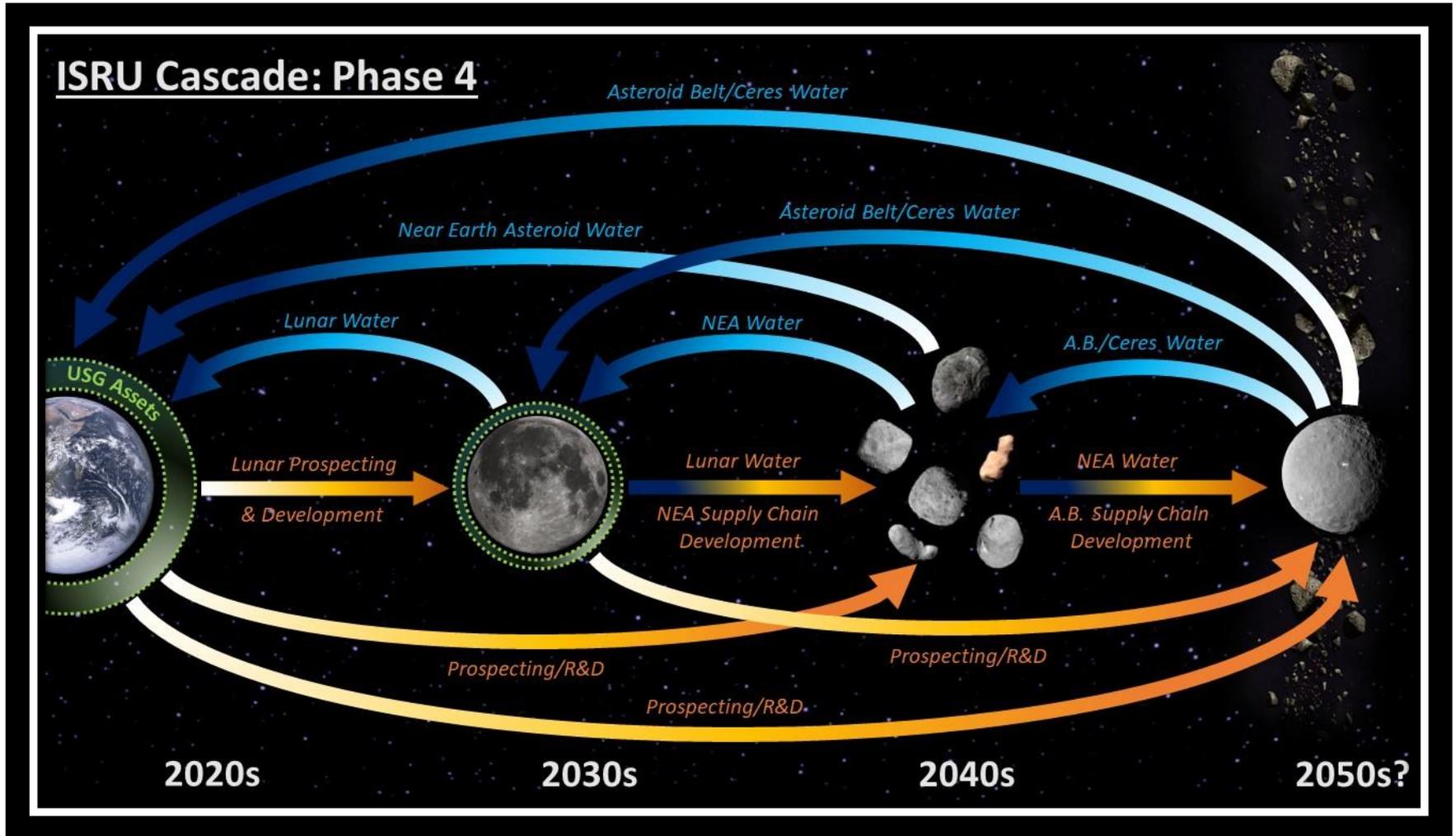
*\*Capped at one million with demand becoming linear thereafter, as # of assets was far too great to be realistic*

# CALCULATING SUPPLY 3: THE ASTEROID BELT

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# SUPPLY CHAIN ROAD MAP



# ASTEROID BELT ASSUMPTIONS AND METHODS

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## Assumptions

- Asteroid belt mass is  $3 \times 10^{21}$  kg
- C-type make up 40% of the population
- C-types are 15% water by mass
- Asteroid intercept requires average delta-V of 8000 km/s
- Fuel losses due to asteroid escape are negligible
- Derived LH2/LOX fuel for Earth return
- About 2/3 of the dry mass of miner is payload
- U.S. Military has 10% share
- Ion/plasma propulsion is used for final fuel delivery

## Methods

- 40% multiplier for percent C-types
- 15% multiplier for percent water
- ~13% multiplier loses due to Earth return
- ~90% multiplier for water loses due to destination orbit insertion
- 10% multiplier for the 10% share of total water

**Leaving  $2.1 \times 10^{18}$  kg for Military Assets**

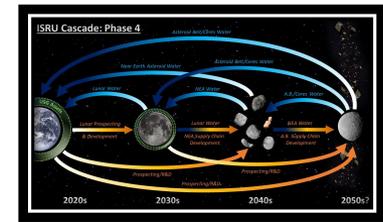
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# ASTEROID BELT RESULTS

Provides inexhaustible resources	<b>Independent Asteroid Belt Supply Chain</b>	
Much of this water is centralized to Ceres	<b>Source Water Mass</b>	1.8 x 10 <sup>20</sup> kg
Could support planned Moon-like manufacturing	<b>Delivered Water Mass</b>	2.1 x 10 <sup>18</sup> kg
Ceres will likely be an important strategic & economic hot spot for humanity	<b>Theoretical Number of Deployable Assets</b>	<b>1 Million*</b>
	<b>Time Until Depletion</b>	<b>2.8 Billion Years*</b>

*\*Capped at one million with demand becoming linear thereafter, as # of assets was far too great to be realistic*

# ALL INDEPENDENT SUPPLY CHAIN OUTCOMES



Given a 10% share of the total resources and estimated requirements for resource return, based on the average military asset with 10X maneuverability and 2% exponential growth, each supply chain could independently support:

	Lunar Supply Chain	Near Term NEA Supply Chain	Long Term NEA Supply Chain	Asteroid Belt Supply Chain
Source Water Mass	$1.2 \times 10^{12}$ kg	$1.6 \times 10^9$ kg	$6.7 \times 10^{14}$ kg	$1.8 \times 10^{20}$ kg
Delivered Water Mass	$5.2 \times 10^{10}$ kg	$5.3 \times 10^7$ kg	$9.8 \times 10^{12}$ kg	$2.1 \times 10^{18}$ kg
Theoretical Number of Deployable Assets	3.4 Million	3,400	1 Million*	1 Million*
Time until Depletion	510 Years	170 Years	12,800 Years*	Unlimited*

\*Capped at 1 million w/ linear demand after for asset replacement, as the exponential # of assets was unrealistic